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LAUNCH REQUIREMENTS- J24 VEHICLE 1610

30 JULY 1965

ATTACHED IS THE LAUNCH REQUIREMENTS AND LIMITATIONS FOR PAYLOAD J24 VEHICLE 1610.

[redacted]
CHIEF
PAYLOAD INTEGRATION

DISTRIBUTION

[redacted]

1024 Gen file

[redacted]

TOP SECRET [redacted]

1.0 SCOPE

1.1 GENERAL.

THE FOLLOWING REQUIREMENTS GOVERN THE CONDITIONS UNDER WHICH THE VEHICLE WILL BE LAUNCHED WITH THE A/P PAYLOAD SYSTEM. ANY DEVIATION FROM THE PRESCRIBED LIMITS SHALL BE CAUSE FOR HOLD. ANY STATUS CHANGES MUST BE REPORTED TO PAYLOAD INTREGRATION IMMEDIATELY AFTER OCCURRENCE FOR EVALUATION. ALL DISCREPANCIES AND DEVIATIONS MUST BE CORRECTED PRIOR TO RESUMPTION OF VEHICLE LAUNCH COUNT-DOWN.

1.2 DESIRED OBJECTIVE

ALL CAMERA OPERATIONS AFTER MATING SHALL BE CALLED 'ALPHA CHECKS' AND SHALL BE MICROWAVED TO STC. VAFB-AP SHALL BE RESPONSIBLE FOR IMPLEMENTING THIS OBJECTIVE.

2.0 PAYLOAD INTERNAL TEMPERATURE.

65 +/- 10 DEG. F. FROM MATING TO T-4 HOURS.
65 +/- 5 DEG. F. FROM T-4 THROUGH LAUNCH.

TEMPERATURE OF PAYLOAD SYSTEM SHALL BE MONITORED AND LOGGED EVERY ONE-HALF HOUR FROM MATING TO LAUNCH

3.0 RELATIVE HUMIDITY.

50 PERCENT OR LESS AT ALL TIMES.

4.0 N2 PRESSURE.

THE N2 MUST BE CONNECTED AND FLOWING WHEN THE SYSTEM IS ON THE LAUNCH PAD. MAXIMUM OFF TIME IS ONE-HALF HOUR PER DAY.

5.0 SRV TRANSMISSION FREQUENCY

RECOVERY TLM	228.2 +/- 0.1 MC
RECOVERY BEACON	235.0 +/- 0.1 MC

6.0 POWER.

POWER MUST BE APPLIED TO THE PAYLOAD INTERFACE WHENEVER THE PAYLOAD IS TRANSPORTED.

7.0 THERMAL BLANKET

THERMAL BLANKET MUST REMAIN ON THE PAYLOAD FROM MATING UNTIL LAUNCH.

8.0 PRIMARY TELEMETRY READOUTS.

PAYLOAD CHECKOUT (MUST BE VERIFIED THROUGH VEHICLE TELEMETRY)

8.1 LENS ROTATION, HORIZON IDLER, AND CENTER OF FORMAT INSTRUMENT 1 CHANNEL 09, LINK I. NO BACK UP CHANNEL IS AVAILABLE.

8.2 LENS ROTATION, HORIZON IDLER, AND CENTER OF FORMAT INSTRUMENT 2 CHANNEL 10, LINK I. NO BACK UP CHANNEL IS AVAILABLE.

8.3 RING A CUMMUTATOR (.4 X 60) CHANNEL 13, LINK I AND POINTS LISTED BELOW. THE PRIMARY COMMUTATED POINTS MUST BE VERIFIED DURING PAYLOAD CONFIDENCE AFTER MATING AND CHECKOUT TASKS RUNS.

MONITOR FUNCTION	VERIFICATION PRIMARY CH-LK-PT	VOLTS	TOL.	ALTERNATE CH-LK-PT	VOLTS	TOL.
INSTR. 1 DOOR EJECT	13-1-04	1.3	+/-.2	NONE	---	---
INTERNAL PRESSURE-CONIC	13-1-07	0.0	+/-.2	NONE	[redacted]	
INSTR. 2 DOOR EJECT	13-1-13	1.3	+/-.2	NONE		
FAIRING SEPARATION	13-1-16	1.3	+/-.2	NONE	---	---
INSTR.1 CYCLE COUNT 1	13-1-22	0.55 STEP	+/-.15	13-1-25	4.05 OR GREATER	4.05 OR GREATER
INSTR.1 CYCLE COUNT 10	13-1-23	0.55 STEP	+/-.15	13-1-25	4.05 OR GREATER	4.05 OR GREATER
INSTR.1 CYCLE COUNT 100	13-1-24	0.55 STEP	+/-.15	13-1-25	4.05 OR GREATER	4.05 OR GREATER
FOOTAGE PUT INSTR. 1	13-1-25	4.05 OR GREATER		8-2-56	4.05 OR GREATER	
INSTR.2 CYCLE COUNT 1	13-1-27	0.55 STEP	+/-.15	13-1-31	0.9 OR LESS	0.9 OR LESS
INSTR.2 CYCLE COUNT 10	13-1-28	0.55 STEP	+/-.15	13-1-31	0.9 OR LESS	0.9 OR LESS
INSTR.2 CYCLE COUNT 100	13-1-29	0.55 STEP	+0-.15	13-1-31	0.9 OR LESS	0.9 OR LESS
CALIBRATE PLUS	13-1-30	5.0	--	13-1-10	5.0	---

FOOTAGE POT INSTR. 2	13-1-31	0.9 OR LESS	0.9 OR LESS	8-2-47	0.9 OR LESS	0.9 OR LESS
SEPARATION MON. SRV 1	13-1-33	0.28	+/-.1	16-1-25	0.28	+/-.1
CONTINUITY LOOP SRV-1	13-1-34	5.38	+/-.2	NONE	---	---
MODE MONITOR REC1/REC2	13-1-41	1.0	+/-.2	NONE	---	---
SEPARATION MON. SRV 2	13-1-43	1.3	+/-.1	NONE	---	---
CONTINUITY LOOP SRV-2	13-1-44	5.38	+/-.2	NONE	---	---
RECOVERY BATTERY SRV-1	13-1-48	0.0	+/-.2	NONE	---	---
RECOVERY BATTERY SRV-2	13-1-50	0.0	+/-.2	NONE	---	---
N2 BOTTLE PRESSURE	13-1-53	3.0 OR GREATER		NONE	---	---
FILM DOOR CLOSURE	13-1-56	4.7	+/-.2	NONE	---	---
CALIBRATE ZERO	13-1-57	0.0	---	13-1-19	0.0	---
SYC. PULSE	13-1-58	5.5	[redacted]			
SYC. PULSE	13-1-58	5.5	+/-.2	[redacted]		

ALL COMMAND SELECTOR POINTS AS LISTED PER LAUNCH REQUIREMENTS LIST IN APPENDIX I.

9.0 HOMING OF CAMERA SCAN ARMS.

9.1 BOTH CAMERA SCAN ARMS MUST BE PROPERLY HOMED PRIOR TO TERMINAL COUNT.

10.0 LAUNCH REQUIREMENTS COMMAND SETTINGS.

10.1 ALL STEPPING SWITCHES MUST BE POSITIONED IN ACCORDANCE WITH THE COMMAND SETTINGS LIST IN APPENDIX I PRIOR TO TERMINAL COUNT.

11.0 FILM CONSUMPTION PRIOR TO LAUNCH.

11.1 MINIMUM FILM CONSUMPTION PRIOR TO LAUNCH SHALL BE 100 CYCLES ON EACH INSTRUMENT.

11.2 SHOULD THE PAYLOAD SYSTEM STAY IN A LOADED CONDITION FOR AN EXTENDED PERIOD, THE SYSTEM SHALL BE OPERATED FOR 10 CYCLES EVERY FOUR DAYS.

12.0 RESPONSIBILITY.

12.1 IT SHALL BE THE RESPONSIBILITY OF THE SENIOR A/P PAYLOAD ENGINEER TO ENSURE THE IMPLEMENTATION OF THE RESTRICTIONS AND REQUIREMENTS LISTED HEREIN. IN ADDITION, HE IS CHARGED WITH THE RESPONSIBILITY OF SUPPLYING THE FOLLOWING SYSTEM INFORMATION IMMEDIATELY PRIOR TO LAUNCH TO FLIGHT OPERATIONS AND COMPUTER SERVICES BY TELEPHONE.

12.1.1 FINAL FLIGHT SYSTEM WEIGHTS

12.1.1.1 TOTAL SYSTEM WEIGHT, IN POUNDS. .....

	A SRV S/N 622	B SRV S/N 643
12.1.1.2 RETRO ROCKET WEIGHT, LBS.	.....	.....
12.1.1.3 PARACHUTE WEIGHT, LBS.	.....	.....

12.1.2 AT THE TIME THE FLIGHT FILM IS FIRST SPLICED TO THE SYSTEM, RECORD THE CYCLE COUNTER READINGS, CASSETTE FOOTAGE POT VOLTAGE AND LENGTH OF TAKE-UP CASSETTE LEADER ON EACH INSTRUMENT.

	MASTER	SLAVE
A. CYCLE COUNTERS, CYCLES	.....	.....
B. FOOTAGE POT VOLTAGE, VOLTS	.....	.....
C. TOTAL SYSTEM LEADER, FT.	.....	.....

12.1.3 IF ANY OFF SPOOLING IS PERFORMED, THE FOLLOWING IS REQUIRED AT THE TIME THE FINAL SPLICE IS MADE BETWEEN THE FLIGHT FILM AND THE TAKE-UP LEADER. RECORD THE CYCLE COUNTER READINGS, CASSETTE FOOTAGE POT VOLTAGES, AND REMAINING LENGTH OF LEADER IN THE TAKE-UP CASSETTES.

	MASTER	SLAVE
A. CYCLE COUNTERS, CYCLES	.....	.....
B. FOOTAGE POT VOLTAGE, VOLTS	.....	.....
C. REMAINING TAKE-UP LEADER, FT.	.....	.....

12.1.4 CYCLE COUNTER AND FILM FOOTAGE POT READINGS FOR EACH INSTRUMENT, AT LAUNCH.

	MASTER	SLAVE
A. CYCLE COUNTERS, CYCLES	.....	.....
B. FOOTAGE POT VOLTAGE, VOLTS	.....	.....

12.1.5 CLOCK ERROR, STATIC RUN, IN MICROSECONDS .....

12.1.6 TOTAL LENGTH OF FLIGHT FILM OFF-SPOOLED FROM EACH SUPPLY SPOOL INCLUDING ALL CONTROL STRIPS AND SAMPLES.

A. MASTER ..... FT.

B. SLAVE ..... FT.

12.1.7 TOTAL LENGTH OF FILM OFF-SPOOLED FROM EACH S/I INSTRUMENT.

A. S/I A STELLAR ..... FT.

A INDEX ..... FT.

B. S/I B STELLAR ..... FT.

B INDEX ..... FT.

APPENDIX I PAYLOAD LAUNCH REQUIREMENT LETTER

LAUNCH REQUIREMENT- COMMAND SETTINGS

THE FOLLOWING COMMAND SETTINGS ARE SPECIFIED FOR THE J24 PAYLOAD 1610 VEHICLE. DATE OF ISSUE .....

NO.	FUNCTION	COMMAND SELECTOR POSITION	VERIFICATION PRIMARY CH-LK-PT	VOLTS	TOL	VERIFICATION SECONDARY CH-LK-PT	VOLTS	TOL
6	V/H RAMP LEVEL	8	13-1-02 13-1-03	2.0 4.0		8-2-24 8-2-26	2.0 4.0	
8	V/H RAMP AMPLITUDE	3	13-1-05 13-1-06	1.0 3.0		8-2-28 8-2-30	1.0 3.0	
9	PROGRAM	4	13-1-08 13-1-09	1.0 4.0		8-2-32 8-2-34	1.0 4.0	
10	V/H RAMP DELAY	6	13-1-11 13-1-12	2.0 2.0	5%	8-2-44 8-2-45	2.0 2.0	5%
11	INSTR. MODE	1	13-1-14 13-1-15	1.0 1.0		8-2-48 8-2-50	1.0 1.0	
12	INTERMIX POSITION	11	13-1-17 13-1-18	4.0 4.0		8-2-52 8-2-53	4.0 4.0	
15	INTERMIX MODE	4	13-1-20	4.0		8-2-55	4.0	

LAUNCH REQUIREMENTS- CAMERA SYSTEM

LOADING MONITORS- THE FOLLOWING REPRESENTATIVES HAVE BEEN DESIGNATED RESPONSIBLE DURING LOADING OF THE FLIGHT SYSTEM

- CUSTOMER- PRIME [redacted]
  ALTERNATE [redacted]
  AP- PRIME [redacted]
  ALTERNATE [redacted]
  BOSTON PRIME [redacted]
  ALTERNATE [redacted]
- OR DESIGNATED REPRESENTATIVE.

THE FOLLOWING SETTINGS/REQUIREMENTS ARE SPECIFIED FOR THE J24 PAYLOAD 1610 VEHICLE.

PANORAMIC LENS SETTINGS-

	INSTRUMENT 1 (MASTER)	INSTRUMENT 2 (SLAVE)
SLIT DIMENSIONS	0.225 inch	0.150 inch
FILTER TYPE	Wratten 25	Wratten 21

NOTE- SLIT DIMENSIONS MUST BE MEASURED PRIOR TO INSTALLATION.

HORIZON OPTICS SETTINGS -

	INSTRUMENT 1 (MASTER)	INSTRUMENT 2 (SLAVE)
SUPPLY HORIZONS-		
APERTURE	F 6.8	F 8.0
SPEED	1/100	1/100
FILTER	Wratten 25	Wratten 25
TAKE-UP HORIZONS-		
APERTURE	F 8.0	F 6.8
SPEED	1/100	1/100
FILTER	Wratten 25	Wratten 25

STELLAR INDEX OPTICS SETTINGS-

	STELLAR INDEX A	STELLAR INDEX B
STELLAR LENS-		
APERTURE	F 1.8	F 1.8
SPEED	2.0 Sec.	2.0 Sec.
FILTER	None	None

INDEX LENS-

APERTURE	F 4.5	F 4.5
SPEED	1/500	1/500
FILTER	Wratten 21	Wratten 21

FILM NOMENCLATURE-

PANORAMIC INSTRUMENTS

	INSTRUMENT 1 (MASTER)	INSTRUMENT 2 (SLAVE)
PRIMARY		
TYPE	7J-40-16000	7J-40-16000
EMUL. DATA	213-5-6-5	213-5-6-5
WT. AND SPCCL NO.	89.7-80.0-5s-82t	89.8-80.2-5s-101B
BOX NO.	28	28
SECONDARY		
TYPE	7J-40-16000	7J-40-16000
EMUL. DATA	222-1-7-5	222-1-7-5
WT. AND SPCCL NO.	88.3-78.9-5s-B59t	88.3-78.9-5s-B60B
BOX NO.	40	40

STELLAR INDEX-

	STELLAR INDEX A		STELLAR INDEX B	
	STELLAR	INDEX	STELLAR	INDEX
PRIMARY-				
TYPE	3J34-75	7J33-135	3J34-75	7J33-135
EMUL. DATA	124-35-8-5	104-14-6-5	124-35-8-5	104-14-6-5
SECONDARY				
TYPE. DATE	3J34-75	7J33-135	3J34-75	7J33-135
EMUL. DATA	124-35-8-5	104-14-6-5	124-35-8-5	104-14-6-5

PANORAMIC OFFSPOOLING REQUIREMENTS-

	MASTER	SLAVE
ORINGINAL LENGTH-FT.	16000	16000
LENGTH OF OFF SPOOL-FT.	0 +/-10	0 +/-10
LENGTH TO LOAD-FT.	16000 less control samples	16000 less control samples

STELLAR INDEX FILM OFFSPOOLING REQUIREMENTS

	STELLAR INDEX A		STELLAR INDEX B	
	STELLAR	INDEX	STELLAR	INDEX
ORGINAL LENGTH-FT.	75	135	75	135
LENGTH TO OFF SPOOL-FT.	29 +1 -0	43 +1 -0	29 +1 -0	43 +1 -0
LENGTH TO LOAD-FT.	46	92	46	92

CYCLE RATIO STELLAR INDEX TO PANORAMIC 1 TO 7

STELLAR BAFFLE TYPE-

	STELLAR A	STELLAR B
	11.500	5.000
LAUNCH WINDOW	2130Z - 230[redacted]Z	

APPROVED BY

[redacted]

OPERATIONS AND ANALYSIS [redacted] DATE....

PROGRAM INTERGRATION [redacted] ATE....

RESIDENT OFFICER [redacted] DATE....

LAST PAGE

LAST PAGE

[redacted]